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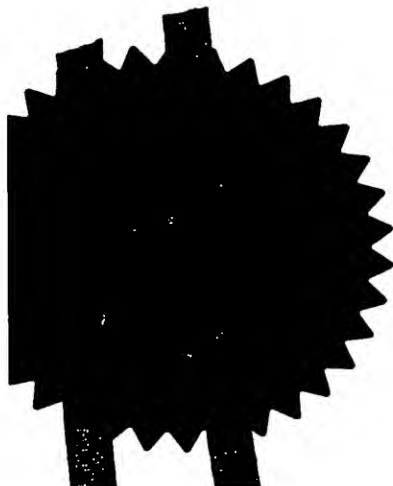
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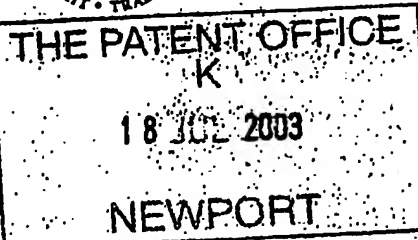
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18 JUL 03 E823627-1 D02481  
P01/7700 0.00-0316863.0

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The Patent Office

Cardiff Road  
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1. Your reference

P34671-NBR/MEA

2. Patent application number

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0316863.0

18 JUL 2003

3. Full name, address and postcode of the or of each applicant (underline all surnames)

Hamilton Erskine Limited  
17 Moss Road, Ballygowan  
Newtownards, Co Down  
BT23 6JQ, Northern Ireland

Patents ADP number (if you know it)

If the applicant is a corporate body, give the country/state of its incorporation

United Kingdom

9676413001

4. Title of the invention

"Improvements Relating to Glass"

5. Name of your agent (if you have one)

"Address for service" in the United Kingdom to which all correspondence should be sent (including the postcode)

Murgitroyd & Company  
Scotland House  
165-169 Scotland Street  
Glasgow  
G5 8PL

Patents ADP number (if you know it)

1198015

6. If you are declaring priority from one or more earlier patent applications, give the country and the date of filing of the or of each of these earlier applications and (if you know it) the or each application number

Country

Priority application number  
(if you know it)

Date of filing  
(day / month / year)

7. If this application is divided or otherwise derived from an earlier UK application, give the number and the filing date of the earlier application

Number of earlier application

Date of filing  
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8. Is a statement of inventorship and of right to grant of a patent required in support of this request? (Answer 'Yes' if:

Yes

- a) any applicant named in part 3 is not an inventor, or
  - b) there is an inventor who is not named as an applicant, or
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# Patents Form 1/77

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Continuation sheets of this form

Description

15

Claim(s)

Abstract

Drawing(s)

6 + 6 92

10. If you are also filing any of the following, state how many against each item.

Priority documents

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Statement of inventorship and right to grant of a patent (Patents Form 7/77)

Request for preliminary examination and search (Patents Form 9/77)

Request for substantive examination (Patents Form 10/77)

Any other documents (please specify)

11. I/We request the grant of a patent on the basis of this application.

Signature

Murgitroyd & Company

Date

17 July 2003

12. Name and daytime telephone number of person to contact in the United Kingdom

Mark Earnshaw

0141 307 8400

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1     Improvements Relating to Glass

2

3     The present invention relates to improved ballistic  
4     and blast- and hurricane-resistant optically  
5     transparent composite materials involving glass.

6

7     There have been many suggestions for "bullet-proof"  
8     and "blast-proof" transparent windows and the like,  
9     either for civilian purposes such as for use in  
10    aircraft, or for military purposes, especially  
11    protection against enemy and terrorist attack.

12    However, with the developing threat from  
13    international terrorism and events such as those of  
14    September 11 2001, many governments and major  
15    organisations are re-appraising their security  
16    requirements. High velocity weapons and better  
17    explosives are increasingly available to terrorists  
18    and the like. Whilst traditional 'bullet-proof'  
19    glass will still be required, there is now an  
20    increasing need for certain key installations,  
21    persons and equipment, especially in and around  
22    military and high governmental locations, to be

1 protected against a higher level of threat than  
2 previously considered necessary.

3  
4 Where pure optical transmission for a window is not  
5 a necessity, there are many available materials  
6 having high strength and impact resistance.  
7 However, where optical transparency of 'normal'  
8 windows and glazing is desired, e.g. for military  
9 base houses and offices, current forms of glazing  
10 are only adequate for protection against low  
11 velocity bullets (e.g. from small arms), and low  
12 levels of blast. Most current forms of 'bullet  
13 proof' glass use several layers of glass bonded by  
14 adhesive polymer film. The energy of the projectile  
15 is dissipated over increasingly large areas of  
16 blast. To some extent the projectile can be  
17 deformed or fragmented and can be deviated from the  
18 original line of attack. The energy is directed  
19 towards a direction different to the previous path,  
20 resulting in further dissipation of energy, e.g. as  
21 shown in figure 1.

22

23 Typical design solutions involve either glass/glass  
24 combinations or glass/polycarbonate (PC)  
25 combination. The latter offer an advantage in that  
26 they are lighter than the former, but they often  
27 have delamination problems. The effect of bonding  
28 of PC to glass is also difficult as PC has a  
29 substantially higher rate of thermal expansion than  
30 glass. This causes high stress levels in the  
31 bonding interlayer during temperature changes which  
32 often leads to delamination.

1 The PC designs are often 'complex', particular as  
 2 the level of protection required increases. The  
 3 number of layers can cause problems with optical  
 4 interference and secondary image formation because  
 5 of the number of glass/PC interfaces. There may  
 6 also be weight or thickness limitations preventing  
 7 their use in particular applications. This is shown  
 8 in the following tables.

Weapon type & Calibre	Class	Design	Thickness (mm)	Weight (kg/m <sup>2</sup> )	Transmittance (%)
Hand Gun 9mm Luger	BR2/C1	6 <sup>2</sup> PC5 <sup>2</sup> 3-12-ESG6	35	47	77
Rifle 0.223 (5.56*45)hc	BR5/C3	8 <sup>2</sup> 6 <sup>2</sup> PC6 <sup>2</sup> 6 <sup>2</sup> PC6 8 <sup>2</sup> PC8 <sup>2</sup> 6-12- 6 <sup>2</sup> PC8 <sup>2</sup> 3-20-ESG6	39 82	71 95	64 ?
Rifle 0.308 (7.62*51)	BR6/C4	8 <sup>2</sup> 8 <sup>2</sup> 6 <sup>2</sup> PC6 <sup>2</sup> 6 <sup>2</sup> PC6 8 <sup>2</sup> PC8 <sup>2</sup> 3-12- 10 <sup>2</sup> PC8 <sup>2</sup> 3-20-ESG6	49 85	93 102	? ?
Rifle 0.308 (7.62*51)hc	BR7/C5	6 <sup>2</sup> 8 <sup>2</sup> 8 <sup>2</sup> PC8-20- 6 <sup>2</sup> 8 <sup>2</sup> 8 <sup>2</sup> PC8	91	143	58

29 US5665450 discusses the introduction of glass fibres  
 30 and glass ribbons into transparent composites, but,  
 31 as it states, the introduction of glass fibres into

1 an optically transparent polymer destroys the  
2 transparency of the polymer.

3  
4 US5665450 considers that the introduction of glass  
5 ribbons provide a higher degree of optical clarity  
6 and lower level of distortion than glass fibres.  
7 However the photographs in US5665450 indicating the  
8 degree of optical clarity of fibre and ribbon-  
9 reinforced materials still show distortion even  
10 based on photographic reproduction of relatively  
11 indistinctive photographs. Figure 7 shows  
12 percentage like transmission as a function of  
13 temperature and wavelength. However, it can be seen  
14 that the percentage transmission barely gets above  
15 80% at the lowest temperature and highest wavelength  
16 measured. The lowest temperature measured is at  
17 30°C, which is also not a temperature generally  
18 encountered in many countries on a regular basis. It  
19 is interesting that the percentage transmission in  
20 US5665450 was not measured at more temperate or  
21 freezing temperatures. Moreover, 80% optical  
22 transmission is very poor in comparison with the  
23 expectancy of 'normal' glass, which should be at  
24 least 90% at all temperatures. It is appreciated  
25 that the human eye can easily recognise or perceive  
26 a less than 100% optical transmission of light  
27 through a 'transparent' material.

28  
29 In essence, there is a requirement for an optically  
30 transparent composite material having about or at  
31 least 90% optical transmission over a range of  
32 temperatures, including below 0°C, and also able to

1 withstand high velocity ballistic projection whilst  
2 having a relatively low manufacturing cost.

3  
4 According to one aspect of the present invention,  
5 there is provided an optically transparent composite  
6 material comprising at least one glass/resin/glass  
7 lamination, wherein the resin is a PRR material  
8 having optical fibre-reinforcement therein.

9  
10 The term "PRR" refers to 'polycarbonate replacement  
11 resins', a range of materials provided by Chemetall  
12 GmbH of Frankfurt, Germany, and generally defined in  
13 their International Patent Application No WO  
14 01/38087A1. The PRR materials are a range of  
15 transparent cast resins that can consist of reactive  
16 acrylate and methacrylate monomers, acrylate and  
17 methacrylate oligomers, bonding agents and  
18 initiators. The content of WO 01/38087A1 defining  
19 these materials is incorporated herein by way of  
20 reference.

21  
22 The term "PRR" also extends to similarly provided  
23 polyurethane resins, often termed "PUR".

24  
25 A range of commonly available PRR materials are sold  
26 under the trade name Naftlolan®. The Naftlolan  
27 materials are provided in a range of different  
28 formulations to provide slightly different  
29 properties. A list of product data of certain  
30 polyurethane Naftlolan materials are listed in  
31 Tables 2 and 3 hereinafter, by way of example only.



1 PRR materials have been found to have several  
2 advantages over previously used polymer glass  
3 lamination layers. Firstly, the refractive index of  
4 PRR material overlaps very closely with many types  
5 of glass. Secondly, PRR materials have been found  
6 to expand and contract at very close rates with that  
7 of glass, thus leading to minimal if ever cracking  
8 or delamination (due to internal stress) during any  
9 thermal expansion and contraction of the composite  
10 material. Thirdly, PRR materials are relatively  
11 very easy to use and set in transparent composite  
12 materials, especially compared with processes of  
13 curing previously used types of polymers and resins.  
14 They are also useable in designs incorporating  
15 complex curves.

16  
17 Because PRR materials have a co-efficient of  
18 expansion and contraction very close to glass, these  
19 materials are usable to provide optically  
20 transparent composite materials with glass over a  
21 much greater range of temperatures than, e.g. that  
22 shown in US5665450. In particular, the present  
23 invention is designed to provide a ballistic-  
24 resistant optically transparent composite material  
25 which is usable at temperatures even as low as  $-15^{\circ}\text{C}$   
26 to  $-40^{\circ}\text{C}$ , generally  $-20^{\circ}\text{C}$ , e.g. the temperature of  
27 windows in military installations in certain  
28 countries such as Canada, as well as temperatures  
29 going up to  $30^{\circ}\text{C}$  to  $40^{\circ}\text{C}$ , such as the temperature of  
30 windows in more tropical countries. To that extent,  
31 the difference in co-efficiency of glass, such as a

1 normal silica-based glass, and PRR materials,  
2 deviates little over a wide temperature range.

3  
4 Table 4 hereinafter lists the refractive indices of  
5 a number of resins, including a number of the  
6 Naftolan range, indicating their close refractive  
7 index to that of glass in general.

8  
9 In general, the refractive index of the PRR  
10 materials are sufficiently close to readily  
11 available types of glass, such as a silica-based  
12 glass, that the optical transparency of the  
13 composite material of the present invention is as  
14 good as that from any current glass/glass or  
15 glass/PC laminations.

16  
17 The fibre reinforcement in the PRR layer of the  
18 composite material of the present invention can be  
19 provided by any know type of "fibre material", being  
20 for instance in the form of filaments, or in the  
21 form of particles such as beads, or even powders, as  
22 long as such fibre material wholly or very  
23 substantially has the same refractive index as glass  
24 across all or most the wavelengths of optical light.  
25 Such glass fibres are well known in the art, one  
26 such available product being sold under the trade  
27 name Tyglas by Fothergill Engineered Fabrics.

28  
29 The fibre reinforcement provide the PRR intermediate  
30 layer with improved strength because of their well  
31 known ability to laterally transmit impact energy.  
32 Meanwhile, PRR materials also have improved utility

1 as a resin to band the fibre material fillers  
2 because of their similar co-efficient of thermal  
3 expansion and adhesive strength to glass.

4  
5 In the present invention, the thickness of the glass  
6 and PRR layers, and the density of fibre  
7 reinforcement in the PRR layer, can vary according  
8 to the qualities of the final composite material  
9 desired. Cost and physical properties are factors  
10 in considering the thickness of the layers. One  
11 known ratio of thickness is glass/PRR/glass of  
12 6/20/4mm; this is provided by way of example only.

13  
14 Indeed, a major facet of PRR material is that its  
15 strength is independent of its thickness. Many  
16 types of resins and adhesives only have strength for  
17 a minimal thickness, as their use is to bond  
18 together the layers (e.g. of glass) on each side,  
19 rather than provide any inherent strength of their  
20 own right. PRR has been found not only to provide  
21 good bonding to glass, but also have internal  
22 strength in its own right. The thickness of the PRR  
23 layer is therefore independent of the thickness of  
24 the glass layers either side.

25  
26 The nature of "high velocity ballistic protection"  
27 can be defined in general terms as the difference  
28 between a hand gun and a rifle, e.g. above a NATO  
29 5.56 or 7.62mm ball.

30  
31 According to a second aspect of the present  
32 invention, there is provided a process for making an

1 optically transparent composite material as herein  
2 before defined, comprising the steps admixing the  
3 PRR material with the optical fibre-reinforcement,  
4 and allowing the combination to cure and set between  
5 the two layers of glass.

6

7 Further information on the curing of PRR resins may  
8 be found in WO 01/38087A1.

9

10 Meanwhile, increasing power and sophistication of  
11 explosive-technology, means that 'blast-proof'  
12 optically transparent material is also desired  
13 having increasing strength. In this regard, it is  
14 now generally desired to provide blast-resistant  
15 optically transparent material having the ability to  
16 withstand a blast of 500kg TNT or equivalent at 40m.

17

18 US patent No 3953630 discloses a laminated  
19 transparent assembly suitable for use as a  
20 windscreen for a high speed vehicle wherein high  
21 strength flexible material is embedded in a plastic  
22 material, laid between two layers of glass. The  
23 flexible material extends beyond the transparent  
24 assembly, so as to be directly conjoined with the  
25 structure of the vehicle. Thus, as any blast causes  
26 deformation of the transparent assembly (as part of  
27 the impact absorption), the high strength flexible  
28 material provides a direct bond between the vehicle  
29 structure bolts and the transparent assembly,  
30 hopefully thereby resisting complete separation of  
31 the two and travel of the transparent assembly into  
32 the vehicle.

1  
2 However, US3953630 only discloses the use of  
3 polyvinylbutyral (PVB) as the plastic layered to  
4 provide the bonding between the glass sheets and the  
5 flexible material. In addition, manufacture of the  
6 transparent assembly in US3953630 requires an  
7 altering of the conventional laminating technique,  
8 in order to provide good bonding between a number of  
9 PVB sheets, and the glass. This requires pre-  
10 heating treatment, insertion of the full assembly  
11 including glass sheets in a closed bag to evacuate  
12 all air, followed by heating in a autoclave with  
13 high pressure. This method of manufacture has not  
14 lent itself to cost-efficient production for a  
15 number of transparent assemblies, other than for the  
16 very special uses such as our aircraft windscreens  
17 as mentioned.

18  
19 Moreover, PVB in particular is a material only  
20 designed to provide good bonding between glass  
21 layers. It is typically only 1-2mm thick. Further  
22 thickness of layer is not desired, as PVB has little  
23 internal strength in its own right.

24  
25 In a third aspect of the present invention, there is  
26 provided a laminated optically transparent assembly  
27 comprising at least one glass/resin/glass  
28 lamination, and having one or more high tensile  
29 strength flexible material reinforcement pieces  
30 extending laterally from the resin layer to provide  
31 increased attachment of the assembly to a surround,  
32 wherein the resin is a PRR material.

1  
2 PRR materials are those as defined herein above. As  
3 well as the greater similarity of refractive index  
4 and co-efficient of thermal expansion of PRR  
5 material to glass, the PRR-flexible material and  
6 PRR-glass bonding has been found to be superior to  
7 that of prior materials such as PVB.  
8

9 Meanwhile, the assembly of the present invention  
10 still provides the degree of flexibility desired for  
11 a blast-resistant window, with the reinforced  
12 attachment of the window to the surround, such as  
13 the window rebate of frame.  
14

15 The high tensile strength flexible material may be  
16 similar to that disclosed in US3953630, i.e. woven  
17 fabric or woven glass fibre material or polyester  
18 fibre material. One such product is Kevlar®.  
19

20 Preferably, the flexible material extends wholly or  
21 substantially around opposites sides of the complete  
22 transparent assembly, to provide flexibility of  
23 attachment to the surround.  
24

25 As for the ballistic-resistant material described  
26 hereinabove, the thickness of the glass and resin  
27 layers of the blast-resistant assembly can follow  
28 those well known in the art. One suitable dimension  
29 for the glass/resin/glass in 4m glass, 4mm PRR and  
30 3mm glass.  
31

1 The thickness of the PRR layer can indeed be up to  
2 40-50mm thick, as PRR has inherent strength  
3 independent of thickness as mentioned above. To  
4 that extent, the PRR material can be as thick and  
5 therefore as strong as desired, as all the strength  
6 from a blast is taken by the resin (whilst any glass  
7 shatters).

8  
9 The ability to provide a PRR layer of any thickness  
10 provides a further benefit.

11  
12 Thus, according to a forth aspect of the present  
13 invention, there is provided a blast-resistant  
14 composite material comprising at least one layer of  
15 PRR material having at least one reinforcement piece  
16 extending wholly or substantially across the PRR  
17 layer.

18  
19 Preferably, the reinforcement piece is a series of  
20 strips of bars or other reinforcement means, more  
21 preferably forming a grid or grid-like structure  
22 wholly or substantially across the composite  
23 material.

24  
25 The PRR material is that as defined hereinabove.  
26 The reinforcement piece can be one or more of woven  
27 rovin, webbing, webbing material or even metal  
28 material. The use of a metallic grid provides the  
29 same effect as a "muntin" system which uses metallic  
30 reinforcement grid alongside a glazing panel, but  
31 not actually therein. The present invention  
32 therefore achieves the same effect and strength as a

1 muntin system, but as a one piece assembly, thereby  
2 significantly reducing assembly and installation.

3  
4 The blast-resistance is achieved because the PRR  
5 layer can be any thickness desired, e.g. up to 40-  
6 50mm, which is able to accommodate reinforcement  
7 pieces, whereas previous resins were not able to  
8 achieve such thickness, and thereby accommodate  
9 reinforcement therein.

10  
11 The benefit of achieving reinforcement within the  
12 PRR material is that each 'section' created by the  
13 reinforcement piece or pieces, e.g. each small  
14 section within the grid, can be regarded as having  
15 its own frame, as thus regarded as a separate  
16 section in terms of analysis against blast. As is  
17 well known in the art, the blast-resistance of a  
18 small section is greater than that of a large  
19 section. By dividing the composite panel into a  
20 number of small sections, significant blast-  
21 resistance is achieved.

22  
23 It is noted that the optical transparency of blast-  
24 resistant panels using the muntin system is not as  
25 important as that described for other aspects of the  
26 present invention, so that the comparative  
27 refractive index is not as important as that as  
28 described above in relation to other aspects of the  
29 present invention.

30  
31 It will be recognised by those skilled in the art  
32 the composite materials and assemblies could also be



1 used to provide hurricane or the like resistance,  
2 and thus the present invention is extended thereto.  
3 Embodiments of the present invention will now be  
4 described by way of example only and with reference  
5 to the accompanying drawings in which:  
6

7 Figure 1 is schematic cross-sectional view of the  
8 impact of a projectile against a current multi-glass  
9 laminated window pane;  
10

11 Figure 2 is a cross-sectional view of a optically  
12 transparent composite material according to one  
13 embodiment of the present invention;  
14

15 Figure 3 is a laminated optically transparent  
16 assembly according to a second embodiment of the  
17 present invention.  
18

19 As previously mentioned, figure 1 shows how the  
20 energy of a projectile is dissipated over  
21 increasingly large areas of glass of a known glass  
22 PC lamination pane, leading to a large area of glass  
23 shattered from the left hand side.  
24

25 Figure 2 shows a optically transparent composite  
26 material 2 comprising a glass/resin/glass  
27 lamination. Within the PRR resin layer 4 are a  
28 series of traditional fibre glass woven rovings 6.  
29

30 To produce the material, the rovings 6 were secured  
31 between two panes of glass 8, and the PRR resin 4  
32 was injected into the cavity. The resin 4 flows up

1 the inside of the glass 8 and disperses through the  
2 woven roving 6, wetting the fibres and forming an  
3 excellent bond.  
4

5 Figure 3 shows a blast-resistant assembly 10 mounted  
6 to a wall 12. Between the two panes of glass 14, a  
7 2 inch wide unidirectional glass fibre woven roving  
8 16 was bonded into the same PRR resin 18 as  
9 mentioned above. The complete assembly 10 was  
10 located in the rebate of a window frame 20, and the  
11 roving reinforcement material 16 fixed to the frame  
12 20 by adhesive, and also by means of a lateral bolt  
13 22.  
14

15 The assembly 10 was tested in a Hannsfield 20k-w  
16 tensometer. Loads in excess of 8000N were applied  
17 before the fibre woven 16 broke. Considerably  
18 greater loads could be achieved with the use of  
19 thicker fibres or different types of fibres.  
20

21 The present invention provides ballistic-resistant  
22 and blast-resistant assemblies providing protection  
23 against much higher levels of protection from high  
24 velocity weapons and explosives than currently known  
25 with current forms of glazing. Production of the  
26 assemblies is also comparatively simple and cost  
27 effective compared to previous types of similar  
28 assemblies, which used less suitable polymers and  
29 plastic material.

1/8

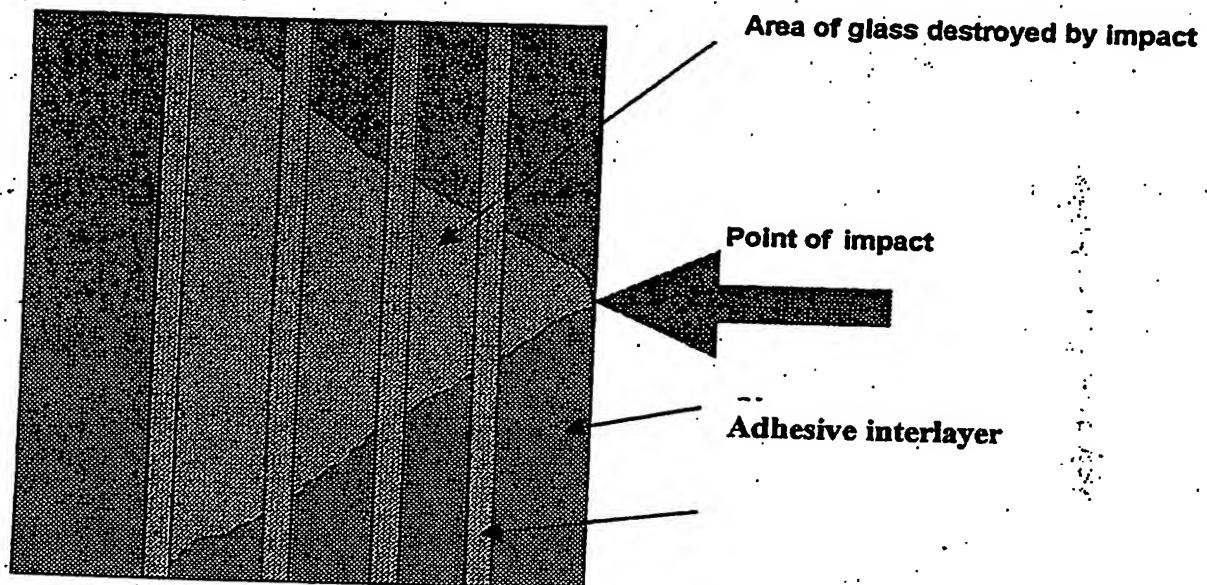


Fig 1

2/6

fig 2

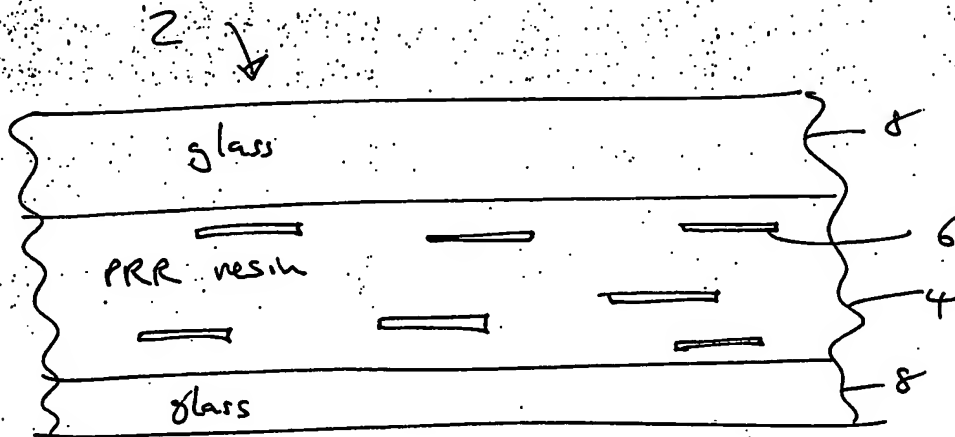
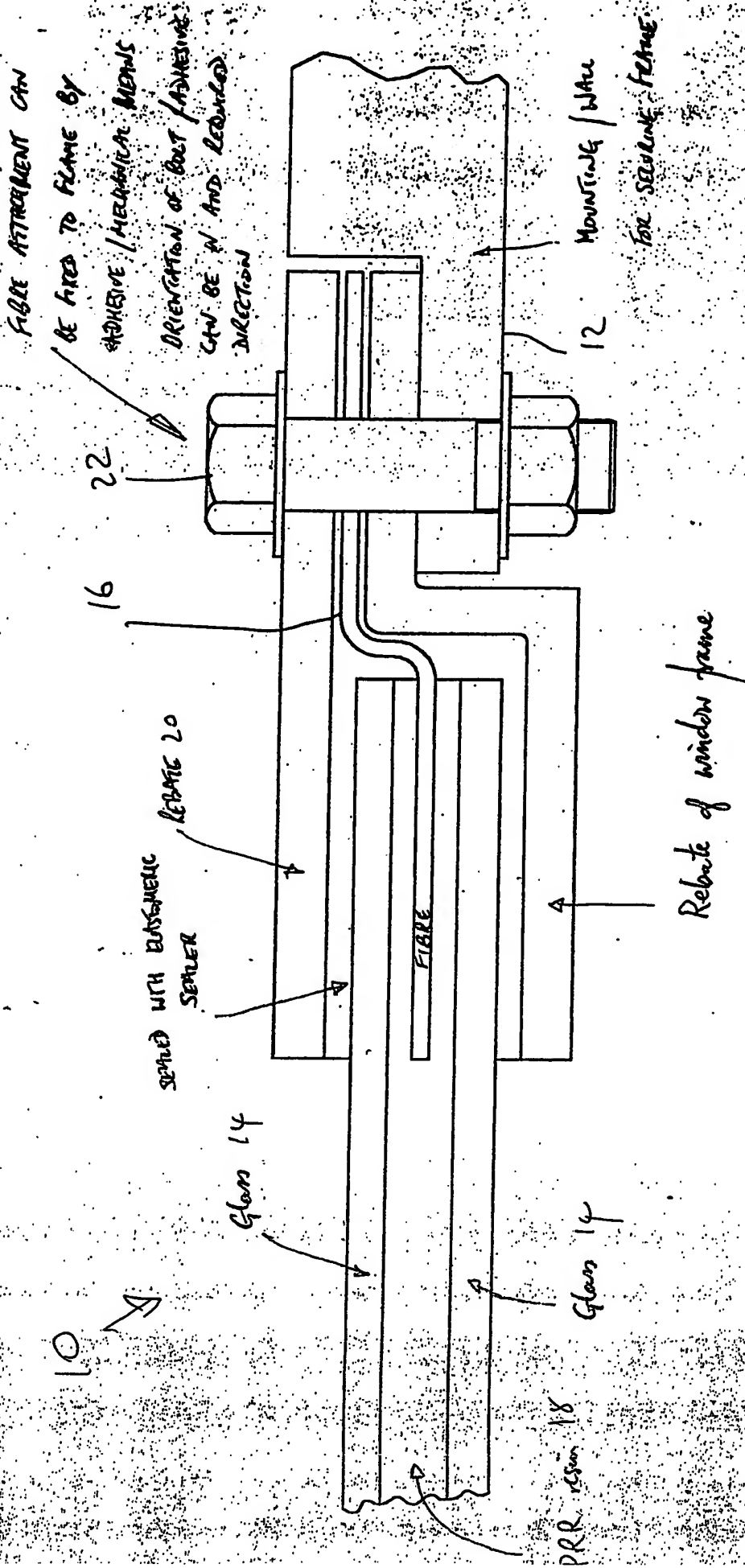


Fig 3



# Product Data of Polyurethane Composite Materials

4/6

Physical Properties	Naftolan PU-A 700 / PU-B 304	Naftolan PU-A 206 / PU-B 606	Naftolan VP-PA 0911 / VP-PB 2110	Naftolan VP-PA 1011 / VP-PB 2110	Naftolan VP-PA 2601 / VP-PB 2110	Naftolan VP-PA 0511 / VP-PB 2610	Standard
Density							
Max							
Min							
Range possible in one layer							
<b>Strength &amp; Stiffness</b>							
Tensile strength [MPa]	16	16	19	18	5	2.1	DIN 53 504
25% module [MPa]	5	1.2	15	6	1.2	0.7	DIN 53 504
50% module [MPa]	6	1.7	14	7	1.8	1.2	DIN 53 504
100% module [MPa]	8	2.5	15.5	10	3.2	1.9	DIN 53 504
Tensile modulus							
Tensile yield point							
Elongation at break [%]	180	360	150	160	150	120	DIN 53 504
Compressive strength							
Compressive modulus							
Compressive yield point							
Flexural strength							
Flexural modulus (also called modulus of elasticity)							
Flexural yield point							DIN EN ISO 527
In-plane shear strength							
In-plane shear modulus							
<b>Toughness</b>							
Impact strength							
Impact strength with loading rate							
<b>Hardness</b>							
Shore A hardness	90	73	85	80	65	55	DIN 53 505
<b>Adhesion to glass</b>							
Bond shear strength							
Compression shear strength [MPa] 4/24 mm glass/ glass	18	7.5	20	20	10	2.5	
<b>Thermal</b>							
Thermal coefficient of expansion							
Thermal conductivity							
Specific heat capacity (at 20 °C)							
<b>Resistance</b>							
Resistance							
Resistivity							
Temperature coefficient of resistance							

These values are for guidance only and do not represent a specification.



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# Product Data of Polyurethan Composite Materials

Physical Properties	Naftolan PU-A 700 / PU-B 304	Naftolan PU-A 206 / PU-B 606	Naftolan VP-PA 0911 / VP-PB 2110	Naftolan VP-PA 1011 / VP-PB 2110	Naftolan VP-PA 2601 / VP-PB 2110	Naftolan VP-PA 0511 / VP-PB 2610	Standard
<b>Flammability</b>							
Melting point							
Boiling point							
Smoke emission							
Critical Oxygen Index (COI) (minimum oxygen fraction in an oxygen-nitrogen mixture which will support steady state combustion of the plastic. Plastics with COI >0.21 are self extinguishing)							
<b>Chemical Resistance</b>							
Acids							
Alkalis							
Organic solvents							
The materials are not resistant against Alkalis The resistance against organic solvents depends very much on the tested solvents.							
<b>Weatherability/degradation</b>							
Ageing - temp cycling test							
UV absorption							
Adsorption of water							
<b>Additional Data</b>							
Viscosity (23°C) [mPa·s]	450	320	600	515	435	360	
Viscosity (23°C) [mPa·s]	3250	85	85	85	85	85	
Viscosity (23°C) [mPa·s]	2550	200	225	200	245	230	
Specific weight (23°C) [g/cm³]	1.02	1.01	1.03	1.02	1.02	1.01	
Specific weight (23°C) [g/cm³]	1.04	1.05	1.05	1.05	1.05	1.05	
Specific weight (23°C) [g/cm³]	1.04	1.037	1.04	1.03	1.03	1.02	
Volume Shrinking [%]	2	2	4	4	3	2	
Processing time (23°C) [min]	45	15	40	30	30	30	
Curing time (23°C) [h]	12	48	12	24	24	12	
Storage time before delivery (18°C to 23 °C) [d]	4	6			4		
Shore A hardness after 1 day (cured resin) DIN	65		75	65	50	40	
Shore A hardness after 7 days (cured resin)	90	73	95	90	65	55	
Thermal conductivity (DIN 52612) [W/(m²K)]	0.18						DIN 52612
Transmittance (DIN 67507) (assembly 6/8/4mm) [%]	90						DIN 67 507
Transmittance (DIN 67507) (assembly 6/12/4mm) [%]	88						DIN 67 507
Transmittance (DIN 67507) (assembly 6/20/4mm) [%]	88						DIN 67 507

These values are for guidance only and do not represent a specification.

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Table 4

# Refractive Index liquid / cured Resins (at 20°C)

Resin	n <sub>D20</sub> liquid	n <sub>D20</sub> cured
UV 11	1,4362	-
UV 22	1,4417	1,4813
UV 33	1,4396	-
ICE-Gießharz EP 1309-103	1,4434	-
UV 203	1,4370	1,4713
S 700 M	1,4299	-
S 696 M	1,4272	-
Naftolan VP-PA 0511	1,4542	-
Naftolan VP-PA 1011	1,4568	1,4844 *
Naftolan PU-A 206	1,4540	-
Naftolan VP-PB 2110	1,4169	-
Naftolan PU-B 606	1,4777	-
Naftolan VP-PA 2601	1,4553	-
Naftolan PU-A 700	-	-
Naftolan PU-B 304	1,4739	-

\* cured with corresponding B component



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